

**Warsaw University
of Technology**



**Faculty of Power and
Aeronautical Engineering**

WARSAW UNIVERSITY OF TECHNOLOGY

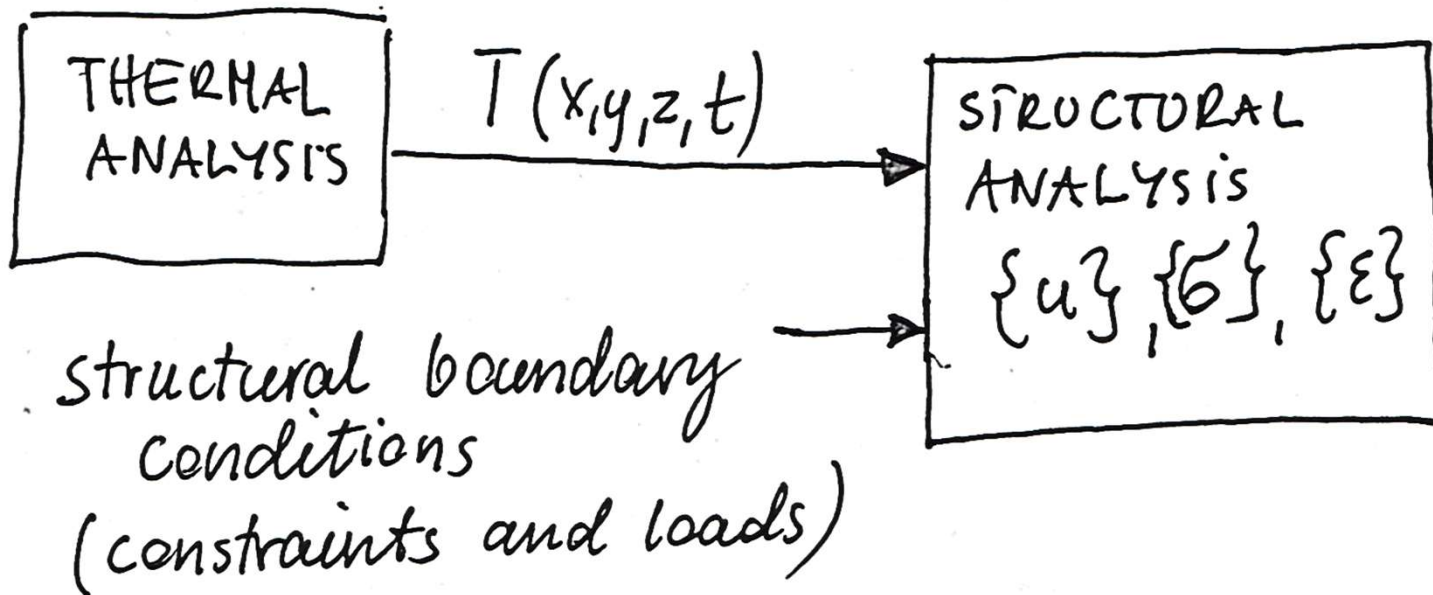
Institute of Aeronautics and Applied Mechanics

Finite element method 2 (FEM 2)

Structural analysis with thermal effect

10.2021

THERMAL STRESS ANALYSIS



STRUCTURAL ANALYSIS WITH THERMAL EFFECT

TOTAL STRAIN:

$$\begin{Bmatrix} \epsilon \end{Bmatrix}_{6 \times 1} = \begin{Bmatrix} \epsilon \end{Bmatrix}_{\tau, 6 \times 1} + \begin{Bmatrix} \epsilon \end{Bmatrix}_{e, 6 \times 1}$$

$$\begin{Bmatrix} \epsilon \end{Bmatrix}_{6 \times 1} = [B]_{6 \times n_e} \cdot \begin{Bmatrix} q \end{Bmatrix}_{n_e \times 1}$$

n_e - no. of degrees of freedom in a finite element

STRESS

$$\begin{Bmatrix} \sigma \end{Bmatrix}_{6 \times 1} = [D]_{6 \times 6} \cdot \begin{Bmatrix} \epsilon \end{Bmatrix}_{e, 6 \times 1} = [D]_{6 \times 6} \cdot \left(\begin{Bmatrix} \epsilon \end{Bmatrix}_{6 \times 1} - \begin{Bmatrix} \epsilon \end{Bmatrix}_{\tau, 6 \times 1} \right) =$$

constitutive matrix

$$= [D]_{6 \times 6} \cdot \left([B]_{6 \times n_e} \cdot \begin{Bmatrix} q \end{Bmatrix}_{n_e \times 1} - \begin{Bmatrix} \epsilon \end{Bmatrix}_{\tau, 6 \times 1} \right)$$

Virtual work principle:

$$\underbrace{[\delta q]_e}_{1 \times n_e} \cdot \underbrace{\{F\}_e}_{n_e \times 1} = \int_{\Omega_e} \underbrace{[\delta \epsilon]}_{1 \times 6} \cdot \underbrace{\{\delta\}}_{6 \times 1} d\Omega_e$$

virtual displacement
nodal forces
virtual strain

$$\underbrace{[\delta q]_e}_{1 \times n_e} \cdot \underbrace{\{F\}_e}_{n_e \times 1} - \int_{\Omega_e} \underbrace{[\delta \epsilon]}_{1 \times 6} \cdot \underbrace{\{\delta\}}_{6 \times 1} d\Omega_e = 0$$

$$\underbrace{[\delta \epsilon]}_{1 \times 6} = \underbrace{[\delta q]_e}_{1 \times n_e} \cdot \underbrace{[B]^T}_{n_e \times 6}$$

hence:

$$\underbrace{[\delta q]_e}_{1 \times n_e} \left(\underbrace{\{F\}_e}_{n_e \times 1} - \int_{\Omega_e} \underbrace{[B]^T}_{n_e \times 6} \cdot \underbrace{\{\delta\}}_{6 \times 1} d\Omega_e \right) = 0$$

= 0

$$\{F\}_e = \int_{\Omega_e} [B]^T \cdot \{\sigma\} d\Omega_e = \int_{\Omega_e} [B]^T \cdot [D] \left([B] \cdot \{q\}_e - \{\epsilon\}_T \right) d\Omega_e$$

$n_e \times 1$ \int_{Ω_e} $n_e \times 6$ 6×1 \int_{Ω_e} $n_e \times 6$ 6×6 $\left(\begin{matrix} n_e \times 6 & n_e \times 1 \\ 6 \times n_e & 6 \times 1 \end{matrix} \right)$

$$= \underbrace{\int_{\Omega_e} [B]^T [D] \cdot [B] d\Omega_e}_{[k]_e} \cdot \{q\}_e - \underbrace{\int_{\Omega_e} [B]^T [D] \cdot \{\epsilon\}_T d\Omega_e}_{\{F_T\}_e}$$

\int_{Ω_e} $n_e \times 6$ 6×6 $6 \times n_e$ $n_e \times 1$ \int_{Ω_e} $n_e \times 6$ 6×6 6×1

$$[k]_e$$

$n_e \times n_e$

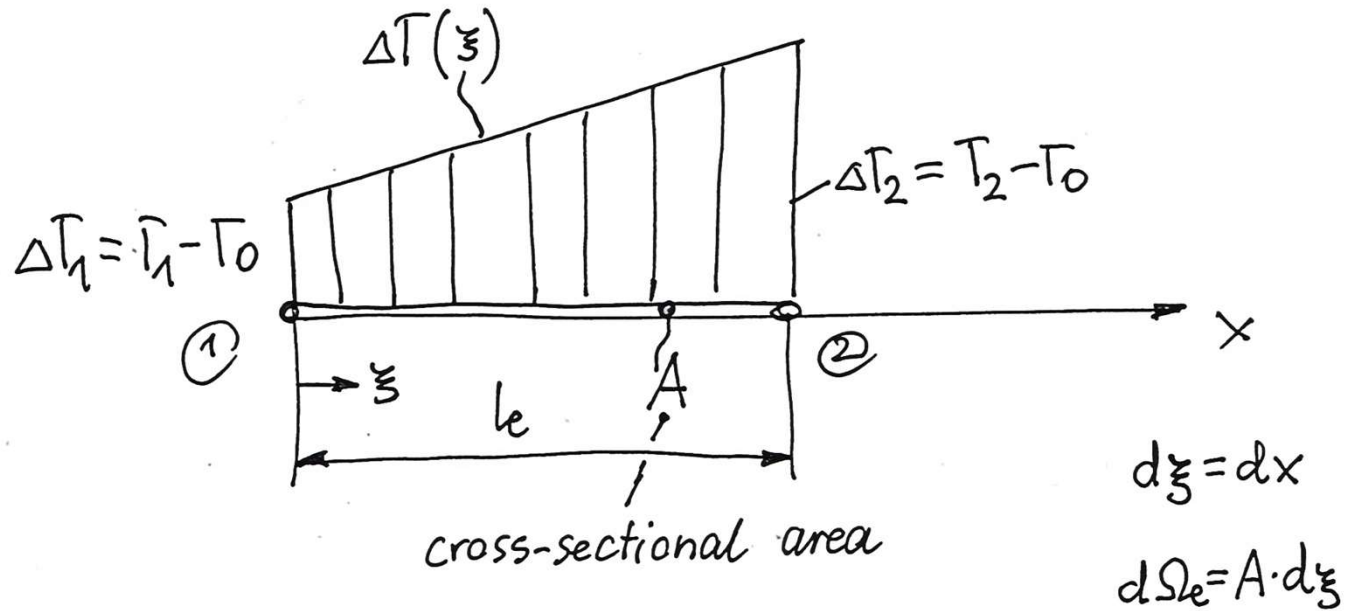
$$\{F_T\}_e$$

$n_e \times 1$

local stiffness matrix

vector of thermal load

EXAMPLE : FIND THERMAL LOAD VECTOR FOR A BAR
FINITE ELEMENT.



$\Delta T_1, \Delta T_2$ - nodal parameters in thermal analysis

$$\Delta T(\xi) = [N_1(\xi), N_2(\xi)] \cdot \begin{Bmatrix} \Delta T_1 \\ \Delta T_2 \end{Bmatrix}$$

$$N_1(\xi) = 1 - \frac{\xi}{l_e}, \quad N_2(\xi) = \frac{\xi}{l_e}$$

$$\epsilon_T(\xi) = \alpha_{se} \cdot \Delta T(\xi) = \alpha_{se} \cdot [N_1, N_2] \cdot \begin{Bmatrix} \Delta T_1 \\ \Delta T_2 \end{Bmatrix}$$

$$\text{FOR "3D": } [B] = [R] \cdot [N] \quad ;$$

$6 \times n_e \quad 6 \times 3 \quad 3 \times n_e$

$$\text{FOR "1D": } [B] = \frac{\partial}{\partial \xi} [N_1, N_2] = \left[\frac{\partial N_1}{\partial \xi}, \frac{\partial N_2}{\partial \xi} \right] = \left[-\frac{1}{l_e}, \frac{1}{l_e} \right]$$

$$\begin{Bmatrix} B \end{Bmatrix}_{2 \times 1} = \begin{Bmatrix} -\frac{1}{l_e} \\ \frac{1}{l_e} \end{Bmatrix}$$

$$\Rightarrow \left\{ F_T \right\}_e = \alpha_{se} \cdot E \cdot A \cdot \int_0^l \begin{bmatrix} \frac{(1 - \frac{\xi}{l})}{l} & -\frac{\xi}{l^2} \\ \frac{(1 - \frac{\xi}{l})}{l} & \frac{\xi}{l^2} \end{bmatrix} d\xi \cdot \begin{Bmatrix} \Delta T_1 \\ \Delta T_2 \end{Bmatrix} =$$

$$= \alpha_{se} \cdot E \cdot A \begin{bmatrix} -\frac{1}{2} & -\frac{1}{2} \\ \frac{1}{2} & \frac{1}{2} \end{bmatrix} \cdot \begin{Bmatrix} \Delta T_1 \\ \Delta T_2 \end{Bmatrix} =$$

$$= \frac{\Delta T_1 + \Delta T_2}{2} \alpha_{se} \cdot E \cdot A \cdot \begin{Bmatrix} -1 \\ 1 \end{Bmatrix}$$